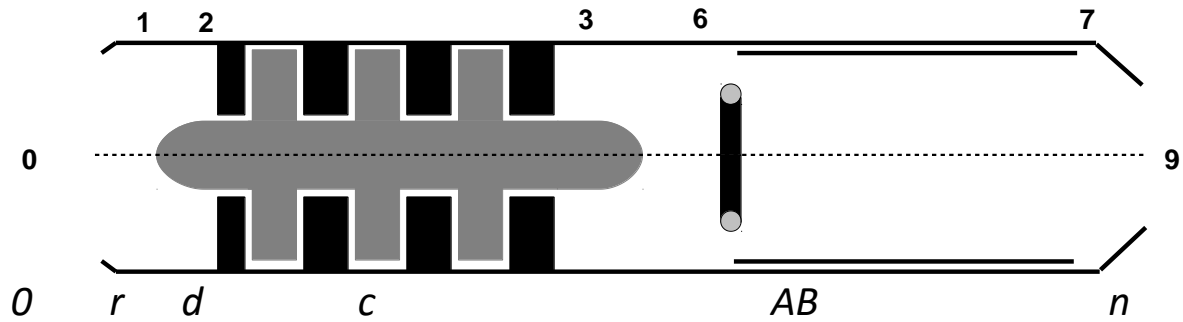


Modelo de Fan Híbrido (Fan elétrico mais pós-queimador = "electric motorjet")



Ar atmosférico:

$$R_c = \frac{\gamma_c - 1}{\gamma_c} c_{pc}$$

$$a_0 = \sqrt{\gamma_c R_c T_0}$$

Gás queimado no pós-queimador:

$$R_{AB} = \frac{\gamma_{AB} - 1}{\gamma_{AB}} c_{pAB}$$

Difusor:

$$V_0 = a_0 M_0$$

$$\tau_r = 1 + \frac{\gamma_c - 1}{2} M_0^2$$

$$\pi_r = \tau_r^{\frac{\gamma_c}{\gamma_c - 1}}$$

$$\pi_d = \pi_{dmax} \eta_r$$

Pós-queimador:

$$\tau_{\lambda AB} = \frac{c_{pAB} T_{t7}}{c_{pc} T_0}$$

Compressor (Electric FAN):

Razão de compressão de cada estágio do compressor (π_{ce}) do FAN e o número de estágios (n_e).

$$\pi_c = \pi_{ce}^{n_e} \qquad \tau_c = \pi_c^{(\gamma_c - 1)/\gamma_c e_c}$$

$$\eta_c = \frac{\pi_c^{(\gamma_c - 1)/\gamma_c} - 1}{\tau_c - 1}$$

Potência elétrica requerida, onde η_e é a eficiência do motor elétrico e seu controlador (*ESC – electronic speed controller*):

$$\dot{W} = \frac{1}{\eta_e} \dot{m}_0 c_{pc} (T_{t3} - T_{t2}) = \frac{1}{\eta_e} \dot{m}_0 c_{pc} T_0 \tau_r \tau_d (\tau_c - 1)$$

Tubeira, na seção de saída:

$$\frac{P_{t9}}{P_9} = \frac{P_0}{P_9} \pi_r \pi_d \pi_c \pi_{AB} \pi_n$$

$$c_{p9} = c_{pAB} \qquad R_9 = R_{AB} \qquad \gamma_9 = \gamma_{AB}$$

$$\dot{m}_9 c_{pAB} T_{t7} - \dot{m}_0 c_{pc} T_{t3} = \eta_{AB} \dot{m}_f h_{PR}$$

$$\frac{(\dot{m}_0 + \dot{m}_f) c_{pAB} T_{t7}}{\dot{m}_0} - \frac{T_{t3}}{T_0} = \frac{\dot{m}_f \eta_{AB} h_{PR}}{\dot{m}_0 c_{pc} T_0}$$

$$f_{AB} = \frac{\dot{m}_f}{\dot{m}_0} = \frac{\tau_{\lambda AB} - \tau_r \tau_d \tau_c}{\frac{\eta_{AB} h_{PR}}{c_{pc} T_0} - \tau_{\lambda AB}}$$

$$\frac{T_9}{T_0} = \frac{T_{t7}/T_0}{T_{t9}/T_9} = \frac{T_{t7}/T_0}{\left(\frac{P_{t9}}{P_9}\right)^{(\gamma_9 - 1)/\gamma_9}}$$

$$M_9 = \sqrt{\frac{2}{\gamma_9 - 1} \left[\left(\frac{P_{t9}}{P_9} \right)^{(\gamma_9 - 1)/\gamma_9} - 1 \right]}$$

$$\frac{V_9}{a_0} = M_9 \sqrt{\frac{\gamma_9 R_9 T_9}{\gamma_c R_c T_0}}$$

$$F = (\dot{m}_0 + \dot{m}_f)V_9 - \dot{m}_0 V_0 + A_9(P_9 - P_0)$$

$$F = a_0 \left[(1 + f_{AB}) \frac{V_9}{a_0} - M_0 + \frac{A_9 P_9}{\dot{m}_0 a_0} \left(1 - \frac{P_0}{P_9} \right) \right]$$

$$\begin{aligned} \frac{A_9 P_9}{\dot{m}_0 a_0} &= (1 + f_{AB}) \frac{A_9 P_9}{\dot{m}_9 a_0} = (1 + f_{AB}) \frac{A_9 \rho_9 R_9 T_9}{\rho_9 V_9 A_9 a_0} \\ &= (1 + f_{AB}) \frac{R_9 T_9}{V_9 a_0} = (1 + f_{AB}) \frac{R_9 T_9}{V_9 a_0^2 / a_0} \\ &= \frac{(1 + f_{AB}) R_9 T_9 / T_0}{\gamma_c R_c V_9 / a_0} \end{aligned}$$

$$\frac{F}{\dot{m}_0} = a_0 \left[(1 + f_{AB}) \frac{V_9}{a_0} - M_0 + (1 + f_{AB}) \frac{R_9 T_9 / T_0 (1 - P_0 / P_9)}{R_c V_9 / a_0 \gamma_c} \right]$$

$$S = \frac{f_{AB}}{F / \dot{m}_0}$$

$$\eta_P = \frac{2V_0(F/\dot{m}_0)}{a_0^2[(1 + f_{AB})(V_9/a_0)^2 - M_0^2]}$$

$$\eta_T = \frac{a_0^2[(1 + f_{AB})(V_9/a_0)^2 - M_0^2]}{2f_{AB}h_{PR}}$$

$$\eta_0 = \eta_P \eta_T$$

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Para determinação da *Vazão Mássica de Saída* utilizando-se o parâmetro de fluxo de massa e sabendo a área de saída (A_9):

$$MFP_9 = \frac{M_9 \sqrt{\gamma_9 / R_9}}{\{1 + [(\gamma_9 - 1)/2] M_9^2\}^{(\gamma_9 + 1)/[2(\gamma_9 - 1)']}}$$

$$\dot{m}_9 = MFP_9 \frac{A_9 P_{t9}}{\sqrt{T_{t9}}}$$

$$\dot{m}_9 = \dot{m}_0 (1 + f_{AB}) \Rightarrow \dot{m}_0 = \frac{\dot{m}_9}{(1 + f_{AB})}$$